

# I. Heat transfer

## Symbology

Symbol	Quantity	Unit
$\lambda, k$	Thermal conductivity	W/(m · K)
$\dot{q}$	Heat flux	W/m <sup>2</sup>
$\dot{Q}$	Heat flow	W
$\alpha$	Heat transfer coefficient	W/(m <sup>2</sup> · K)
$a$	Thermal diffusivity	m <sup>2</sup> /s
$\nu$	Kinematic viscosity	m <sup>2</sup> /s

## 1. Types of heat transfer

### 1.1 Heat conduction

Occurs with all materials when there is a temperature gradient.

#### Steady-state conduction (1st Fourier's law):

$$\dot{q} = -\lambda \frac{dT}{dx}$$

#### Transient conduction (2nd Fourier's law 1D):

$$\frac{\partial T}{\partial t} = \frac{\lambda}{\rho c_p} \frac{\partial^2 T}{\partial x^2}$$

### 1.2 Convective heat transfer

This occurs in moved fluids.

#### Newton's approach:

$$\dot{q} = \alpha (T_1 - T_2) = \frac{\lambda_{\text{fluid}}}{\delta_{\text{th}}} (T_1 - T_2)$$

### 1.3 Radiation

Bodies with a temperature above 0 K emit thermal radiation.

## 2. Heat conduction and conductivity

### 2.1 Heat flux

$$\dot{q} = -\lambda \frac{dT}{dx} = k\Delta T = \frac{\lambda}{\delta} \Delta T$$

Procedure:

$$\vec{q} \int_0^{\delta} dx = -\lambda \int_{T_1}^{T_2} dT \implies \dot{q} = \frac{\lambda}{\delta} (T_1 - T_2)$$

### 2.2 Heat flow

$$\dot{Q} = \frac{\Delta Q}{\Delta t} = \dot{q}A = kA\Delta T = \frac{\lambda}{\delta} A\Delta T$$

## 2.3 Thermal conduction $k^{-1}$ and Thermal conductivity $k$

$$k_{\text{tot}}^{-1} = \sum_{i=1}^n \frac{\delta_i}{\lambda_i} \implies k_{\text{tot}} = \frac{1}{k_{\text{tot}}^{-1}}$$

### 3. Heat conductivity through a hollow cylinder

$$\dot{Q} = -\lambda A \frac{dT}{dr} = -\lambda 2\pi r l \frac{dT}{dr}$$

$$\dot{Q} = 2\pi l \cdot \frac{T_0 - T_1}{\frac{1}{\lambda} \ln \frac{r_1}{r_0}} = 2\pi r_0 l k \Delta T$$

With multilayered walls:

$$\dot{Q} = 2\pi l \cdot \frac{T_0 - T_n}{\sum_{i=1}^n \frac{1}{\lambda_i} \ln \frac{r_i}{r_{i-1}}} = \frac{2\pi r_0 l k_0 (T_0 - T_n)}{\sum_{i=1}^n \frac{1}{\lambda_i} \ln \frac{r_i}{r_{i-1}}}$$

### 3.1 Thermal conductivity $k$

$$k_0 = \frac{1}{r_0 \left( \left( \frac{1}{\lambda_1} \ln \frac{r_1}{r_0} \right) + \left( \frac{1}{\lambda_2} \ln \frac{r_2}{r_1} \right) + \dots + \left( \frac{1}{\lambda_n} \ln \frac{r_n}{r_{n-1}} \right) \right)}$$

$$k_0^{-1} = r_0 \sum_n \left( \frac{1}{\lambda_n} \ln \frac{r_n}{r_{n-1}} \right)$$

### 3.2 Thin-walled pipe approximated as a flat wall

$$\dot{Q}_T = \lambda A \frac{T_0 - T_1}{\delta} ; A = 2\pi l \left( r_0 + \frac{\delta}{2} \right)$$

$$\frac{\dot{Q}_T}{\dot{Q}_W} = \frac{\delta}{\left( r_0 + \frac{\delta}{2} \right) \ln \left( 1 + \frac{\delta}{r_0} \right)} = \frac{\delta}{r_0} \frac{1}{\left( 1 + \frac{\delta}{2r_0} \right) \ln \left( 1 + \frac{\delta}{r_0} \right)}$$

## 4. Convective heat transfer

### 4.1 Reynold's number

It compares inertial to viscous forces.

$$Re = \rho c \frac{L_{\text{char}}}{\eta} = c \frac{L_{\text{char}}}{\nu} \quad \begin{array}{l} Re < 2300: \text{Laminar flow;} \\ Re \approx 2300: \text{Transition zone;} \\ Re > 2300: \text{Turbulent flow} \end{array}$$

### 4.2 Prandtl number

It compares momentum diffusivity to thermal diffusivity. Sets velocity's relative thickness and thermal boundary layers.

$$Pr = \frac{\nu}{a} = \frac{\delta_{\text{fluid}}}{\delta_{\text{th}}} = \frac{\eta c_p}{\lambda} ; a = \frac{\lambda}{\rho c_p} ; \nu = \frac{\eta}{\rho}$$

## 4.3 Nusselt number

Measures convective heat transfer relative to pure conduction across a boundary layer and the heat transfer coefficient  $\alpha$ .

$$Nu = \frac{\alpha \cdot L_{\text{char}}}{\lambda_{\text{fl}}} = \dot{q} \frac{L_{\text{char}}}{\lambda \Delta T_{\text{char}}}$$

$$\alpha = \frac{Nu \cdot \lambda}{L_{\text{char}}}$$

### 4.3.1 Forced convection

$$Nu = c \cdot Re^m \cdot Pr^n$$

where  $c$  is a constant.

For laminar flow over a flat plate:

$$Nu = 0.664 \cdot \sqrt{Re} \cdot \sqrt[3]{Pr}$$

For a flow in the transition zone (Tz):

$$Nu_{Tz} = \sqrt{Nu_{\text{lamin}}^2 + Nu_{\text{turb}}^2}$$

### 4.4 Grashof number

It compares buoyancy to viscous forces.

$$Gr = g\beta\Delta T \frac{L^3}{\nu^2} ; g = 9.81 \text{ N/kg}$$

$$\beta\Delta T = \frac{\rho_{\infty} - \rho_0}{\rho_{\infty}} = \frac{\Delta\rho}{\rho_{\infty}}$$

$$Gr = g \frac{\rho_{\infty} - \rho_0}{\rho_{\infty}} \frac{L^3}{\nu^2}$$

### 4.4.1 Natural convection

#### Peclet number

It compares advective to diffusive heat transport, i.e., the strength of convection relative to thermal diffusion.

$$Pe = RePr$$

#### Rayleigh number

It combines buoyancy driving and thermal diffusion, and it measures the intensity of natural convection.

$$Ra = GrPr$$

## 5. Heat exchanger

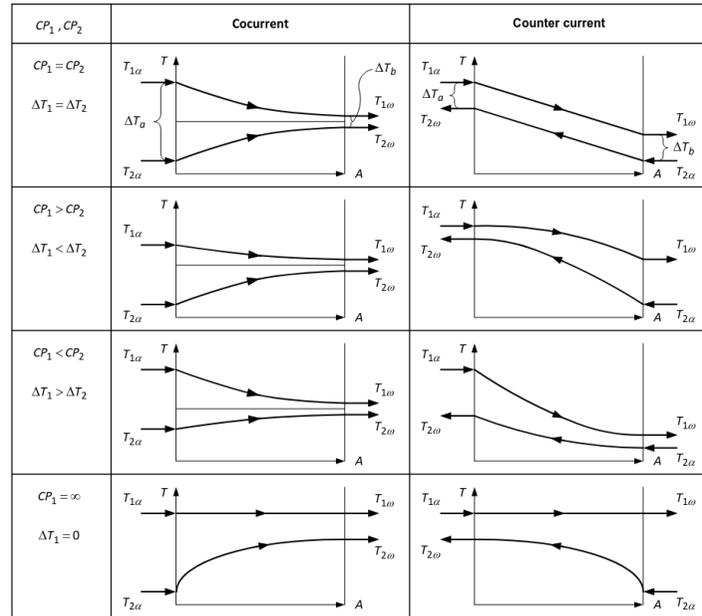
### 5.1 Energy balance

$$\dot{Q} = \dot{m} c_p \Delta\vartheta = CP\Delta\vartheta$$

$$\dot{Q}_1 = \dot{Q}_2$$

$$\dot{m}_1 c_{p1} \Delta\vartheta_1 = \dot{m}_2 c_{p2} \Delta\vartheta_2 \iff CP_1 \Delta\vartheta_1 = CP_2 \Delta\vartheta_2$$

## 5.2 Average temperature difference



### 5.2.1 Logarithmic mean temperature difference (LMTD)

$$\Delta T_m = \frac{\Delta T_a - \Delta T_b}{\ln \frac{\Delta T_a}{\Delta T_b}}$$

where:

$$\Delta T_a = |\vartheta_{1,a} - \vartheta_{2,a}| \quad ; \quad \Delta T_b = |\vartheta_{1,b} - \vartheta_{2,b}|$$

### 5.2.2 Mean temperature heat flow

$$\dot{Q} = kA\Delta T_m$$

### 5.2.3 Heat capacity rate ratio (Capacity ration)

It measures the relative ability of stream 1 vs. stream 2 to carry heat per kelvin:

$$R_1 = \frac{\Delta \vartheta_1}{\Delta \vartheta_2} = \frac{CP_1}{CP_2}$$

## 5.3 Cocurrent vs countercurrent flow heat exchanger

$$\dot{Q} = kA_{co}\Delta T_{m,co} = kA_{count}\Delta T_{m,count}$$

$$\frac{A_{co}}{A_{count}} = \frac{\Delta T_{m,count}}{\Delta T_{m,co}}$$

### 5.3.1 If $\dot{Q}_1 \neq \dot{Q}_2$

$$\frac{A_{co}}{A_{count}} = \frac{\dot{Q}_{count}\Delta T_{m,count}}{\dot{Q}_{co}\Delta T_{m,co}}$$

## II. The 2nd law of thermodynamics

### 6. Recap of the 1st law

#### 6.1 Closed systems

$$q_{12} + w_{12} = u_2 - u_1$$

#### 6.2 Open systems

$$q_{12} + w_{12} = h_2 - h_1 + \frac{1}{2}(c_2^2 - c_1^2) + g(z_2 - z_1)$$

$$w_{t12} = \int v dp + \frac{1}{2}(c_2^2 - c_1^2) + g(z_2 - z_1) + j_{12}$$

$$q_{12} + \int v dp + j_{12} = h_2 - h_1 \stackrel{\text{ideal gas}}{=} c_p(T_2 - T_1)$$

### 7. The 2nd law

#### 7.1 Statements

$$\text{Energy} = \text{Exergy} + \text{Anergy}$$

Processes	Reversible	Irreversible
Energy	Conserved	Reduced → Loss
Entropy	Constant	Increases → Production

#### 7.2 Entropy

Entropy  $S$ : extensive state variable [J/K]

Specific entropy  $s$ : intensive state variable [J/kgK]

Entropy flow  $\dot{S}$ : entropy over time [W/K]

### 8. Entropy balance equation

$$dS = dS_Q + dS_m + dS_{irr}$$

#### 8.1 Entropy balance terms

• Heat transfer across the system boundary:

$$dS_Q = \frac{dQ}{T}$$

• Mass flow across the system boundary:

$$dS_m = \dot{S} = \dot{m} s$$

• Irreversible processes inside the system:

$$\dot{S}_{irr}(t) \begin{cases} dS_{irr} > 0 & \text{irreversible processes} \\ dS_{irr} = 0 & \text{reversible processes} \\ dS_{irr} < 0 & \text{impossible} \end{cases}$$

#### 8.2 Entropy balance equation for closed systems

$$dS = dS_Q + dS_m + dS_{irr}$$

$$dS = \dot{S}_Q dt + \dot{S}_{irr} dt = \left( \frac{\dot{Q}}{T} + \dot{S}_{irr} \right) dt$$

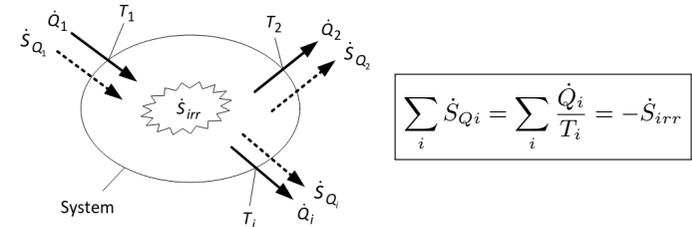
### 8.2.1 Closed system entropy flow balance equation

$$\frac{dS}{dt} = \sum_i \dot{S}_{Qi} + \dot{S}_{irr} \quad ; \quad \dot{S}_{irr} \geq 0$$

For steady-state case:

$$\dot{S}_{irr} = - \sum_i \dot{S}_{Qi} = - \sum_i \frac{\dot{Q}_i}{T_i} \geq 0$$

### 8.2.2 Visual representation



### 8.3 Entropy balance equation for open systems

$$\frac{dS}{dt} = \sum_{\alpha} \dot{m}_{\alpha} s_{\alpha} - \sum_{\omega} \dot{m}_{\omega} s_{\omega} + \sum \dot{S}_Q(t) + \dot{S}_{irr}(t)$$

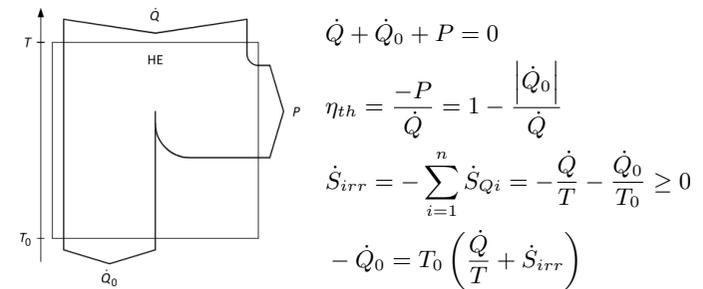
#### 8.3.1 Steady-state flow processes

$$\dot{S}_{irr} = \sum_{\omega} \dot{m}_{\omega} s_{\omega} - \sum_{\alpha} \dot{m}_{\alpha} s_{\alpha} - \sum \dot{S}_Q \geq 0$$

#### 8.3.2 Steady-state and adiabatic flow processes

$$\dot{S}_{irr} = \sum_{\omega} \dot{m}_{\omega} s_{\omega} - \sum_{\alpha} \dot{m}_{\alpha} s_{\alpha} \geq 0$$

### 9. The thermal engine



#### 9.1 Carnot efficiency

$$\eta_{th} = 1 - \frac{T_0}{T} - \frac{T_0 \dot{S}_{irr}}{\dot{Q}} \quad \text{0, ideal}$$

$$\eta_{th,max} = \eta_C(T, T_0) = 1 - \frac{T_0}{T}$$